## STABILITY OF A LAMINAR BOUNDARY LAYER OF A POWER-LAW NON-NEWTONIAN LIQUID

V. V. Zametalin

The stability of a laminar boundary layer of a power-law non-Newtonian fluid is studied. The validity of the Squire theorem on the possibility of reducing the flow stability problem for a power-law fluid relative to three-dimensional disturbances to a problem with two-dimensional disturbances is demonstrated. A numerical method of integrating the generalized Orr-Sommerfeld equation is constructed on the basis of previously proposed [1] transformations. Stability characteristics of the boundary layer on a longitudinally streamlined semiinfinite plate are considered.

In this work, we will study the stability of a laminar boundary layer of fluids with a power rheological law, for which the relation between the stress tensor  $\tau_{ii}$  and strain rate tensor  $\dot{e}_{ii}$  has the form

$$\tau_{ij} = -\delta_{ij}p + k \left| \frac{1}{2} \dot{e}_{ml} \dot{e}_{lm} \right|^{\frac{n-1}{2}} \dot{e}_{ij}, \qquad (1)$$

where i, j=1, 2, 3; k is the consistency index; n is the non-Newtonian index; p is pressure; and  $\delta_{ij}$  is the Kronecker symbol. We will assume that media corresponding to n > 1 are dilating, while media with values n < 1 are pseudoplastic, the latter including, in particular, aqueous solutions of high polymers. The case n=1 corresponds to a Newtonian fluid.

The flow stability of a power-law non-Newtonian fluid has been considered for a plane channel in [2]. The stability of the boundary layer of a power-law fluid was studied in [3, 4] using asymptotic methods. The critical Reynolds numbers for  $0.2 \le n \le 2$  was estimated in [3] using an approximate formula derived there. Neutral stability curves for dilating fluids have been constructed in [4]. Some results [3, 4] are contradictory in the range of values n > 1. This leads to the necessity of a more careful determination of the stability characteristics, which can be attained by numerically integrating the stability equations. In the current work, the stability of the boundary layer of a power-law fluid is studied numerically on the basis of a previous method [1].

Well-known differential motion equations of a power-law non-Newtonian fluid [5] are obtained by substituting Eq. (1) in a stressed deformable continuum equation. Let us represent a nonsteady disturbed flow as the sum of two flows, namely, a steady main flow and small disturbing flow. As usual [6], we will assume that the main flow is laminar and that the components of disturbed motion can be represented in the form

 $\begin{aligned} \widetilde{u} &= u(y) e^{i(\alpha x + \beta z) - i\alpha ct}; \\ \widetilde{v} &= v(y) e^{i(\alpha x + \beta z) - i\alpha ct}; \\ \widetilde{w} &= w(y) e^{i(\alpha x + \beta z) - i\alpha ct}; \\ \widetilde{p} &= p(y) e^{i(\alpha x + \beta z) - i\alpha ct}, \end{aligned}$ 

where  $\alpha$  and  $\beta$  are real values and  $c = c_r + ic_i$  is complex.

The disturbing motion is assumed to be three-dimensional, since it has been proved [7] that the Squire theorem does not hold in the general case for non-Newtonian fluids.

Novosibirsk. Translated from Zhurnal Prikladnoi Mekhaniki i Tekhnicheskoi Fiziki, No. 1, pp. 101-106, January-February, 1976. Original article submitted November 29, 1974.

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We linearize the dimensionless differential equations for total flow, subtracting from them the main flow equations, obtaining

$$i\alpha \operatorname{Reu}(U-c) + vU'\operatorname{Re} + i\alpha \operatorname{Rep} = (U')^{n-1} [nu'' - (\alpha^2 + \beta^2)u + (n-1)i\alpha v'] + (n-1)n(U')^{n-2}U''(u' + i\alpha v);$$
(2)  

$$i\alpha \operatorname{Rev}(U-c) + \operatorname{Rep}' = (U')^{n-1} [v'' - (\alpha^2 + \beta^2)v + i\alpha(n-1)(u' + i\alpha v)] + 2(n-1)(U')^{n-2}U''v';$$
(2)  

$$i\alpha \operatorname{Rew}(U-c) + i\beta \operatorname{Rep} = (U')^{n-1} [w'' - (\alpha^2 + \beta^2)w] + (n-1)(U')^{n-2}U''(w' + i\beta v); i(\alpha u + \beta w) + v' = 0,$$

where U(y) is main flow rate,  $\operatorname{Re} = \rho \operatorname{Uchar}^{2-n} \operatorname{Lchar}^{n} / k$  is the generalized Reynolds number ( $\rho$  is fluid density). The primes denote differentiation with respect to the dimensionless tranverse coordinate y.

Thus, we have four equations to determine the four variables u, v, w, and p in the case of a threedimensional disturbing motion.

Equations (2), after the transformations

$$\overline{\alpha^2} = \alpha^2 + \beta^2$$
,  $\overline{p} \ \overline{R}e = p \operatorname{Re}$ ,  $\overline{\alpha} \overline{R}e = \alpha \operatorname{Re}$ ,  $\overline{v} = v$ ,  
 $\overline{c} = c$ ,  $\overline{\alpha u} = \alpha u + \beta w$ ,

take the form

$$i\overline{\alpha}\overline{\operatorname{Reu}}(U-\overline{c}) + \overline{v}U'\overline{\operatorname{Re}} + i\overline{\alpha}\overline{\operatorname{Rep}} = (U')^{n-1}[n\overline{u''} - \overline{\alpha^2 u} + (n-1)i\overline{\alpha v'}] + (n-1)n(U')^{n-2}U''(\overline{u'} + i\overline{\alpha v});$$
(3)  
$$i\overline{\alpha}\overline{\operatorname{Rev}}(U-\overline{c}) + \overline{\operatorname{Rep}}' = (U')^{n-1}[\overline{v''} - \overline{\alpha^2 v} + i\overline{\alpha}(n-1)(\overline{u'} + i\overline{\alpha v})] + 2(n-1)(U')^{n-2}U''\overline{v'}; i\overline{\alpha u} + \overline{v'} = 0.$$

Equations (3) correspond to two-dimensional disturbing motion with Reynolds number Re less than Re in Eqs. (2). The Squire theorem for a power-law non-Newtonian fluid therefore holds.

In place of Eqs. (2) we then consider the two-dimensional analog of Eqs. (2), which is obtained if we set w = 0 and  $\beta = 0$ . We introduce a dimensionless stream function of the disturbing motion in the form

$$\psi = \varphi(y) e^{i\alpha(x-\epsilon t)}$$

obtaining, after some algebra, the generalized Orr-Sommerfeld equation for a power-law fluid,

$$(U-c) (\varphi'' - \alpha^{2} \varphi) - U'' \varphi = \frac{(U')^{n-3}}{i \alpha \operatorname{Re}} \{ (U')^{2} n (\varphi^{\mathrm{IV}} - 2\alpha^{2} \varphi'' + \alpha^{4} \varphi) + (n-1) [2nU'U'' \varphi'' + [4\alpha^{2} (U')^{2} + nU'U''' + n (n-2) (U'')^{2}] \varphi'' + 2 (n-2) \alpha^{2} U'U'' \varphi' + n\alpha^{2} [U'U''' + (n-2) (U'')^{2}] \varphi \} \}.$$

$$(4)$$

When n=1, the equation turns into the ordinary Orr-Sommerfeld equation for a Newtonian fluid. The generalized Reynolds number for the case of a boundary layer is written in the form

$$\operatorname{Re} = \rho U_0^2 - n \delta^n / k,$$

where  $U_0$  is the free-stream flow rate and  $\delta$  is layer thickness.

The boundary conditions have the form

$$\varphi(0) = \varphi'(0) = 0,$$
 (5)  
 $\varphi(\infty) = \varphi'(\infty) = 0.$ 

Transformations [1] proposed for the ordinary Orr-Sommerfeld equation were used in solving the problem (4), (5) for the eigenvalues.

We define the function  $D_i$  (i =1, 2, 3, 4, 5, 6) by the equations

$$\begin{aligned} D_1 &= \begin{vmatrix} \phi_1 & \phi_2 \\ \phi_1' & \phi_2' \end{vmatrix}, \quad D_2 = \begin{vmatrix} \phi_1' & \phi_2' \\ \phi_1' & \phi_2' \end{vmatrix}, \quad D_3 = \begin{vmatrix} \phi_1' & \phi_2' \\ \phi_1'' & \phi_2' \end{vmatrix}, \\ D_4 &= \begin{vmatrix} \phi_1 & \phi_2 \\ \phi_1'' & \phi_2' \end{vmatrix}, \quad D_5 = \begin{vmatrix} \phi_1' & \phi_2' \\ \phi_1'' & \phi_2' \end{vmatrix}, \quad D_6 = \begin{vmatrix} \phi_1 & \phi_2 \\ \phi_1'' & \phi_2' \end{vmatrix}, \end{aligned}$$

where  $\varphi_1$  and  $\varphi_2$  are the two partial solutions of Eqs. (4) satisfying conditions (5). Then Eq. (4) can be reduced to a system of six ordinary differential equations:

$$\begin{split} D_1' &= D_6 \quad D_4' = D_5 + AD_4 + GD_1 + BD_6, \\ D_2' &= D_5, \quad D_5' = D_3 + AD_5 + BD_2 - ED_1, \\ D_3' &= AD_3 - GD_2 - ED_6, \qquad D_6' = D_2 + D_4, \end{split}$$

where

$$A = 2(1 - n)U''(U')^{-1};$$

$$B = (1 - n)\left[\frac{4\alpha^2}{n} + (n - 2)(U'')^2(U')^{-2} + U'''(U')^{-1}\right] + 2\alpha^2 + \frac{i\alpha}{n}\operatorname{Re}(U - c)(U')^{1 - n};$$

$$G = \frac{2}{n}(1 - n)(n - 2)\alpha^2U''(U')^{-1};$$

$$E = (1 - n)\alpha^2[U'''(U')^{-1} + (n - 2)(U'')^2(U')^{-2}] - \alpha^4 - \frac{i\alpha}{n}\operatorname{Re}(U')^{1 - n}[(U - c)\alpha^2 + U''].$$

We obtain the simple condition  $D_1(0) = 0$  to find the eigenvalues. We carry out the normalization  $Z_i = D_i/D_6$ and eliminate  $Z_5$  by integrating the system  $Z_5 = Z_1Z_3 + Z_2Z_6$ , finally being left with

$$Z_{1} = 1 - Z_{1}(Z_{2} + Z_{4}),$$

$$Z_{2} = Z_{1}Z_{3} - Z_{2}^{2},$$

$$Z_{3}' = AZ_{3} - GZ_{2} - E - Z_{3}(Z_{2} + Z_{4}),$$

$$Z_{4}' = B + GZ_{1} + AZ_{4} + Z_{1}Z_{3} - Z_{4}^{2}.$$
(6)

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TABLE 1

No. of curve from Fig. 5	С	n	k
1	0,6	0,506	0,736
2	0,3	0,546	0,307
3	0,2	0,579	0,173
4	0,1	0,666	0,051
5	0,05	0,789	0,010
6	0	1	0,001

Since the influence of viscosity is absent and, consequently, so are the non-Newtonian properties outside the boundary layer, the boundary conditions at infinity (5) can be carried over to the external edge of the boundary layer (y=1) in the usual fashion [4, 6]. We will write the boundary conditions in the form

 $\varphi''(1)+(1+\alpha)\varphi'(1)+\alpha\varphi(1)=0;$  $\varphi'''(1)+\alpha\varphi''(1)=0$ 

for convenience in calculating values of  $Z_i(1)$ . We set

$$\varphi_1 = 1; \ \varphi_1' = -\alpha; \ \varphi_1' = \alpha^2; \ \varphi_1''' = -\alpha^3;$$
  
 $\varphi_2 = 0; \ \varphi_2' = 1; \ \varphi_2'' = -1 - \alpha; \ \varphi_2''' = \alpha(1 + \alpha).$ 

when y=1, obtaining the boundary conditions for the system (5),

$$Z_1 = -\frac{1}{1+\alpha}; \ Z_2 = -\frac{\alpha}{1+\alpha}; \ Z_3 = 0; \ Z_4 = -\alpha.$$
(7)

Thus, the problem has been reduced to a solution of the system (6) with the boundary conditions (7). The condition  $Z_1(0) = 0$  can be made to hold by varying  $\alpha$ , Re, and c and the eigenvalues are thereby found.

A self-consistent solution of the boundary-layer equations of a power-law fluid for the case of plane longitudinal streamline of a semiinfinite plate was used to determine the velocity profile. The boundaryvalue problem for the ordinary differential equation

$$|F''|^{n-1}F''' + FF'' = 0,$$
  
 $F(0) = 0, F'(0) = 0, F'(\infty) = 1$ 

was numerically solved using the method of group transformations set forth in [8]. The calculated velocities reasonably agree with those presented in [5].

Stability characteristics of the boundary layer of a power-law fluid for the range of values of the non-Newtonian factor  $0.1 \le n \le 1$ , 2, i.e., basically for pseudoplastic fluids of most interest from the practical point of view, were calculated based on this technique. Neutral stability curves are depicted in Figs. 1 and 2 ( $\alpha^* = \alpha \delta^* / \delta$ ; Re\* = Re $\delta^* / \delta$ ; where  $\delta^*$  is displacement thickness).

The dependence of the generalized critical Reynolds number  $\text{Re}^*$  on the parameter n is depicted in Fig. 3 (curve 1), in which the monotonically increasing nature of this function is maintained as we pass through  $n \doteq 1$ . Thus the results qualitatively agree with the previous [3] data obtained asymptotically (curve 2). Qualitatively satisfactory coincidence is observed when  $0.6 \le n \le 1$ . On the other hand, an extremely substantial divergence of the curves occurs in the region of low n. Curves of equally increasing disturbances calculated for the boundary layer of a power-law fluid when n = 0.5 (solid curves) and a Newtonian fluid (broken curves) for identical  $c_i$  are depicted in Fig. 4.

The coordinates of points at which flow loses stability in the boundary layer as a function of the free-stream flow rate  $U_0$  for aqueous solutions of the high-molecular-weight polymer ET-597 were calculated based on our results, using previous [9] data. The family of curves constructed for the different concentrations (Fig. 5) allows the stability of the boundary layer of non-Newtonian power-law and Newtonian fluids to be graphically compared. Concentration C in percent, the non-Newtonian factor n, and consistency index k[Hc<sup>n</sup>/m<sup>2</sup>] for the corresponding curves in Fig. 5 are depicted in Table 1. The value of k is replaced by the viscosity of pure water at 20°C when C=0 and n=1. It is clear in Fig. 5 that, in spite of the general decreasing tendency for stability with increasing non-Newtonian properties, a region exists for low  $U_0$  where the calculated points at which stability is lost for the polymer solutions is situated downstream from that of pure water. Our results lead us to conclude that the non-Newtonian viscosity of high-polymer solutions exerts a destabilizing influence in the case of flow into the boundary layer.

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## CONVERGENT SHOCK WAVE IN AN IDEAL ELASTIC

## NONHOMOGENEOUS MEDIUM

I. V. Simonov

UDC 539.374

The boundary-value problem for symmetric focusing of a shock wave in a medium with variable density under a constant load (model of a porous body with variable initial velocity) is solved. The solution asymptotic is studied. Focusing in a homogeneous medium has been previously studied [1]. One inverse problem related to the choice of the optimal pressure conditions is examined. Constraints on the applicability of the model are touched on.

Suppose a uniform load  $p_0(t)$  is applied to the surface of a sphere (cylinder, layer) whose initial density is a differentiable function of the radius  $[\rho = \rho(r)]$  at a moment of time t = 0. We assume that the load instantaneously attains a finite value  $p_0(t) > 0$  and does not increase any further (the physical meaning of this condition is that of an explosion on the surface); the medium is ideal (without tangential stresses). The density of the medium at any point  $\rho_1$  is set equal to a constant ( $0 < \rho < \rho_1$ ) and remains constant if the pressure at this point reaches values arbitrarily greater than zero. This highly simplified model approximately describes the behavior of a body with variable porosity and uniform skeleton at high loads.

A shock wave will propagate from the surface to the center. The focusing process for the shock wave in a homogeneous medium has been studied in [1]. The purpose of the current report is to investigate the influence of nonhomogeneity on the motion of the medium behind the front of a convergent shock wave. In particular, the variation in the degree of cumulation of a shock wave is of some interest. It may be expected that, as in the case of an ideal gas of variable density [2], the choise of  $\rho(\mathbf{r})$  can either weaken or intensify accumulation.

The following motion and continuity equations hold within the region bounded by the moving surface  $r = R_1(t)$  and the shock wave front r = R(t):

Moscow. Translated from Zhurnal Prikladnoi Mekhaniki i Tekhnicheskoi Fiziki, No. 1, pp. 107-111, January-February, 1976. Original article submitted January 30, 1975.

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